

## 1- Publications in Ship Structural Analysis and Design (1969-2002)

- 1- "Effect of Variation of Ship Section Parameters on Shear Flow Distribution, Maximum Shear Stresses and Shear Carrying Capacity Due to Longitudinal Vertical Shear Forces", European Shipbuilding, Vol. 18. (Norway-1969), Shama, M. A.,
- 2- "Effect of Ship Section Scantlings and Transverse Position of Longitudinal Bulkheads on Shear Stress Distribution and Shear Carrying Capacity of Main Hull Girder", Intern. Shipb. Progress, Vol. 16, No. 184, (Holland-1969), Shama, M. A.,
- 3- "On the Optimization of Shear Carrying Material of Large Tankers", SNAME, J.S.R, March. (USA-1971), Shama, M. A.,
- 4- "An Investigation into Ship Hull Girder Deflection", Bull. of the Faculty of Engineering, Alexandria University, Vol. XII., (Egypt-1972), Shama, M. A.,
- 5- "Effective breadth of Face Plates for Fabricated Sections", Shipp. World & Shipbuilders, August, (UK-1972), Shama, M. A.,
- 6- "Calculation of Sectorial Properties, Shear Centre and Warping Constant of Open Sections", Bull., Of the Faculty of Eng., Alexandria University, Vol. XIII, (Egypt-1974), Shama, M. A.
- 7- "A simplified Procedure for Calculating Torsion Stresses in Container Ships", J. Research and Consultation Centre, AMTA, (EGYPT-1975), Shama, M. A.
- 8- "Structural Capability of Bulk Carriers under Shear Loading", Bull., Of the Faculty of Engineering, Alexandria University, Vol. XIII, (EGYPT-1975), Also, Shipbuilding Symposium, Rostock University, Sept. (Germany-1975), Shama, M. A.,
- 9- "Shear Stresses in Bulk Carriers Due to Shear Loading", J.S.R., SNAME, Sept. (USA-1975) Shama, M. A.,
- 10- "Analysis of Shear Stresses in Bulk Carriers", Computers and Structures, Vol.6. (USA-1976) Shama, M. A.,
- 11- "Stress Analysis and Design of Fabricated Asymmetrical Sections", Schiffstechnik, Sept., (Germany-1976), Shama, M. A.,
- 12- "Flexural Warping Stresses in Asymmetrical Sections" PRADS77, Oct., Tokyo, (Japan-1977), Intern. Conf/ on Practical Design in Shipbuilding, Shama, M. A.,
- 13- "Rationalization of Longitudinal Material of Bulk Carriers, Tehno-Ocean'88, (Jpan-1988), Tokyo, International Symposium, Vol. II, A. F. Omar and M. A. Shama,
- 14- "Wave Forces on Space Frame Structure", AEJ, April, (Egypt-1992), Sharaki, M., Shama, M. A., and Elwani. M.,
- 15- "Response of Space Frame Structures Due to Wave Forces", AEJ, Oct., (Egypt-1992). Sharaki, M., Shama, M. A., and Elwani. M. H.
- 16- "Ultimate Strength and Load carrying Capacity of a Telescopic Crane Boom", AEJ, Vol.41., (Egypt-2002), Shama, M. A. and Abdel-Nasser, Y.

SHIP HULL

**EFFECT OF VARIATION OF SHIP SECTION PARAMETERS ON SHEAR FLOW DISTRIBUTION, MAXIMUM SHEAR STRESSES AND SHEAR CARRYING CAPACITY DUE TO LONGITUDINAL VERTICAL SHEAR FORCE**

By

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*Summary*

The paper examines the effect of varying the different ship section parameters on the shear flow distribution and maximum shear stresses at the section neutral axis due to longitudinal vertical shear force. The contribution of the centre line longitudinal bulkhead to the shear carrying capacity of the main hull girder is also investigated.

The analysis is limited to small tankers having one centre line longitudinal bulkhead, and is carried out in the form of a parameteric study on the University IBM 1620 digital computer. The results are given in terms of ship depth, thickness of side shell plating and the applied shear force and are represented in tabular and graphical forms.

It is concluded that the thickness of longitudinal bulkhead plating and the ratio of the sectional area of the shear carrying members to the total sectional area of the hull longitudinal material are the most important parameters which affect the shear flow distribution, maximum shear stresses, and shear carrying capacity of the centre line longitudinal bulkhead and side shell plating.

Further, it is concluded that it is possible to check graphically the shear carrying capacity of a ship section under the action of a known shear force.

*List of Symbols*

- t = thickness
- t<sub>B</sub> = effective thickness of bottom plating
- t<sub>D</sub> = « « « deck plating
- t<sub>S</sub> = « « « side shell plating
- t<sub>C</sub> = « « « centre line longitudinal bulkhead
- B = breadth
- D = depth
- I = second moment of area of ship section about its own neutral axis
- γ = B/D
- x = t<sub>D</sub>/t<sub>B</sub>
- y = t<sub>C</sub>/t<sub>S</sub>
- z = t<sub>B</sub>/t<sub>S</sub>
- γ, x, y, z, are independent parameters
- β = the normalised distance of the neutral axis from the base line

- ψ = coefficient of midship section second moment of area ( $I = \psi D^3 t_s$ )
- R<sub>W</sub> = ratio of sectional area of shear carrying members (side shell plating + longitudinal bulkhead) to the total sectional area of the hull longitudinal material
- R<sub>L</sub> = ratio of sectional area of centre line longitudinal bulkhead to the total sectional area of the hull longitudinal material
- R<sub>W</sub> and R<sub>L</sub> are dependent parameters
- Q = first moment of area about neutral axis  
=  $\sum A_p \cdot \bar{y}$
- A<sub>P</sub> = sectional area of plating
- $\bar{y}$  = height of centroid of area above neutral axis
- Q<sub>S</sub> =  $\varphi_s D^2 t_s$
- φ<sub>C</sub> = coefficient of first moment of area for the centreline longitudinal bulkhead
- w = φ/ψ
- $\frac{q}{\bar{q}}$  = shear flow in tons/cm  
= mean shear flow
- q<sub>ij</sub> = shear flow at point i in direction ij
- (q)<sub>C</sub> = correcting shear flow
- q<sub>i</sub> = shear flow at point i
- q<sub>r</sub> = resultant shear flow
- (q<sub>C</sub>)<sub>m</sub> = mean shear flow for the centre line longitudinal bulkhead
- (q<sub>S</sub>)<sub>m</sub> = mean shear flow for side shell plating
- (q<sub>C</sub>)<sub>y</sub> = shear flow at a depth y from the neutral axis for the centre line longitudinal bulkhead
- (q<sub>S</sub>)<sub>y</sub> = shear flow at a depth y from the neutral axis for the side shell plating
- η<sub>C</sub> = maximum shear stress for the centre line longitudinal bulkhead
- η<sub>S</sub> = maximum shear stress for the side shell plating
- F<sub>H</sub> = horizontal longitudinal shear force
- F = vertical longitudinal shear force
- F<sub>C</sub> = shear force carried by the centre line longitudinal bulkhead  
= k<sub>C</sub> · F
- k<sub>S</sub> = coefficient of the shear force carried by one side shell plating

- G = modulus of rigidity
- $\zeta_s$  &  $\zeta_c$  = coefficients of the maximum shear stress
- $\eta_s$  &  $\eta_c$
- $\zeta_c = w_c/y$
- $\zeta_s = w_s$

**Method of Calculation**

The method of calculation used in this analysis is given in references [1, 2, 3] and is based on the idea that the structure is allowed to distort under an assumed set of shear forces. A correction is then introduced to satisfy the existing conditions of geometry. The resultant shear flow distribution must satisfy the following conditions of equilibrium [2]:

- a. sum of horizontal shear forces must be zero
- b. sum of vertical shear forces must equal the resultant shear force F
- c. the angle of twist must be zero for each cell and for the whole structure since it is assumed that no torsional moments are applied.

Condition (a) above is valid since we are dealing with a structure subjected only to longitudinal vertical shear force.

Condition (b) implies that the shear force carried by the centre line longitudinal bulkhead and the two sides of a ship must equal the applied longitudinal vertical shear force, i.e.:

$$F = F_c + 2 F_s \dots (1)$$

In this analysis it is assumed that F is uniformly distributed in the transverse direction.

Condition (c) is satisfied, for each cell, by applying an opposite uniform shear flow so that the angle of twist  $\theta$ , resulting from the assumed shear flow distribution, vanishes. This angle of twist is given by:

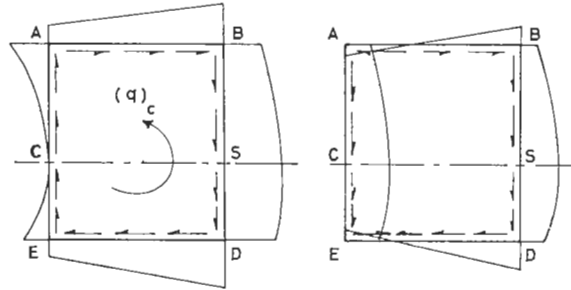
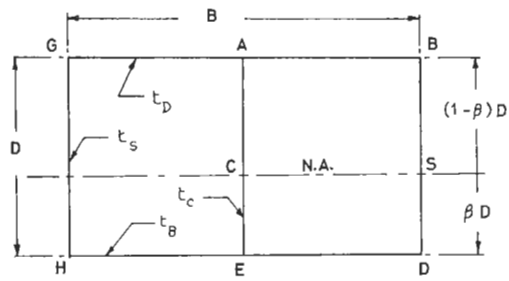
$$\theta = \frac{1}{2AG} \cdot \oint q \frac{\Delta s}{t} \dots (2)$$

where:  $\Delta s$  is taken round the cell

A = sectional area of the cell under consideration.

Using the above three conditions of equilibrium, the shear flow distribution, in a simplified box shape of small tankers having one centre line longitudinal bulkhead, is calculated as follows:

1. The shear flow at point c, see fig. (1), is assumed zero i.e.  $q_c = \text{zero}$
2. At joint A, the shear flow must satisfy the continuity equation i.e.
 
$$q_{AC} = q_{AB} + q_{BC} \dots (4)$$
 But  $q_{AB} = q_{AC}$  because of symmetry  
 Hence  $q_{AC} = 2q_{BC}$



ASSUMED SHEAR FLOW      RESULTANT SHEAR FLOW

Fig. 1.

3. A shear flow distribution is assumed as follows:

$$q_i = \frac{F}{I} \cdot Q_i \dots (5)$$

where:  $Q_i$  = first moment of the area above point i about the neutral axis of the section and is calculated in terms of ship depth, thickness of side shell plating and the different independent parameters as given in Appendix (1).

i.e.  $Q_i = f(B, D, t_s, t_B, t_D, t_c)$

$I$  = the second moment of area about section neutral axis and is calculated in terms of ship depth, thickness of side shell plating and the different independent parameters as given in Appendix (1).

i.e.  $I = F(B, D, t_s, t_B, t_D, t_c)$

Substituting for  $Q_i$  and  $I$  in terms of  $\gamma, x, y$  and  $z$ , we get:

$$Q_i = \varphi_i D^2 t_s \dots (6)$$

$$I = \psi D^3 t_s \dots (7)$$

where  $\varphi_i$  = non-dimensional coefficient of the first moment of area and depends on the position of the point i and the different independent parameters

$\psi$  = non-dimensional coefficient of the second moment of area and de-

depends on the different independent parameters.

It should be noted that, since the plating thicknesses are relatively small in comparison to the section dimensions, the concept of shear flow is used. It should also be noted that the effect of longitudinals and girders are taken into consideration by replacing the actual thickness of any member by an effective thickness calculated as follows:

$$t_e = t_a + \frac{a}{l}$$

where:  $t_e$  = effective thickness of member

$t_a$  = actual thickness of member

$a$  = total sectional area of all other longitudinal materials fitted to the member

$l$  = length of member.

4. Because of symmetry of structure, the shear flow distribution and the correcting shear flow are calculated only for cell ABDE. Using equation (2), the correcting shear flow is calculated as follows:

$$(q_c) \left\{ \sum \frac{\Delta s}{t} \left[ \begin{matrix} E-D \\ D-B \\ B-A \end{matrix} \right] + 2 \sum \frac{\Delta s}{t} [A-E] \right\} - \oint q \frac{\Delta s}{t} = 0$$

$$\text{i.e. } (q)_c = \frac{\oint q \frac{\Delta s}{t}}{\int \frac{\Delta s}{t} + 2 \int \frac{\Delta s}{t}} \quad \dots (8)$$

The correcting shear flow is calculated in terms of the different independent parameters, see appendix (II), and is given by:

$$(q)_c = (w)_c \cdot \frac{F}{D}$$

where:  $(w)_c$  is a non-dimensional coefficient

5. The resultant shear flow at any point  $i$  in each cell is given by:

$$(q_i)_r = q_i - (q)_c \quad \dots (9)$$

This method of calculation has been applied to small tankers having one centre line longitudinal bulkhead to calculate the maximum values of shear stress and the shear loads carried by the centre line longitudinal bulkhead and side shell plating in addition to the shear flow distribution. The calculations are summarised as follows:

- a) Maximum values of the shear flow occurs at points C & S, see fig. (1), on the neutral axis and are given by:

$$q_c = 2 (q)_c = w_c \cdot \frac{F}{D} \quad \dots (10)$$

$$\text{and } q_s = w_s \cdot \frac{F}{D} \quad \dots (11)$$

$$\text{where } w_s = \varphi_s / \psi \\ w_c = 2 (w)_c$$

$\varphi_c$  and  $\varphi_s$  are calculated in Appendix (II) and are given by:

$$\varphi_c = \frac{[0.5 + 0.25y][\beta^3 + (1-\beta)^3] + \frac{3\gamma y}{16Z} [\beta^2 + \frac{(1-\beta)^2}{x}] + \frac{\gamma Z}{2} [\beta^2 + x(1-\beta)^2] + \frac{\gamma^2}{8}}{\frac{\gamma}{2Z} (1 + \frac{1}{x}) + \frac{2}{y} + 1.0} \quad \dots (12)$$

$$q_s = \frac{\gamma x z}{2}(1-\beta) + 0.25(y+2)(1-\beta)^2 - \varphi_c \quad (13)$$

$$\psi = (2+y) \left[ \frac{1}{12} + (0.5-\beta)^2 \right] + \gamma z [\beta^2 + x(1-\beta)^2] \quad (14)$$

$$\beta = \frac{1 + 0.5y + \gamma x z}{2 + y + \gamma z (1+x)} \quad (15)$$

The definitions of  $\psi$  and  $\beta$  are given in Appendix (I).

- b) Maximum shear stresses occur at points C and S and are given by see fig. (1):

$$\tau_s = \frac{q_s}{\bar{t}_s} = \rho_s \cdot \frac{F}{Dt_s} \quad (16)$$

$$\tau_c = \frac{q_c}{\bar{t}_c} = w_c \cdot \frac{F}{Dt_c} \quad (17)$$

Assuming that  $\bar{t}_c/\bar{t}_s = y$

$$\text{Hence, } \tau_c = \rho_c \cdot \frac{F}{Dt_s}$$

where  $\bar{t}_s$  and  $\bar{t}_c$  are actual thicknesses of side shell plating and centre line longitudinal bulkhead respectively.

- c. Shear load carried by the centre line longitudinal bulkhead is given by:

$$F_c = \int_{-\beta D}^{(1-\beta)D} (q_c)_y \cdot dy = (q_c)_m \cdot D \quad (18)$$

where:  $(q_c)_m$  = mean shear flow for the centre line longitudinal bulkhead

$(q_c)_y$  = shear flow at any depth  $y$  from the neutral axis.

The mean shear flow  $(q_c)_m$  is calculated on the assumption that the shear flow distribution is parabolic across the depth of side shell plating and centre line longitudinal bulkhead and is given by:

$$(q_c)_m = \frac{1}{3} [\beta q_{EC} + (1-\beta)q_{AC} + 2q_c] \cdot \frac{F}{I} \quad (19)$$

Substituting for  $q_{EC}$ ,  $q_{AC}$  and  $q_c$  from Appendix (II), into equation (19), we get:

$$(q_c)_m = \left\{ \frac{y}{6} [\beta^3 + (1-\beta)^3] - 2\varphi_c \right\} \frac{F}{D} \quad (20)$$

$$\text{Hence: } F_c = K_c \cdot F \quad (21)$$

$$\text{where: } K_c = \frac{\frac{y}{6} [\beta^3 + (1-\beta)^3] - 2\varphi_c}{\psi} \quad (22)$$

$F_c$  = shear force carried by the centre line longitudinal bulkhead

It should be realised that the participation of the centre line longitudinal bulkhead to the shear carrying capacity will only be fulfilled when the bulkhead is adequately stiffened against instability (4).

- d. Shear force carried by the side shell plating is calculated from the equilibrium condition given by equation (1) i.e.:

$$\begin{aligned} F_s &= (F - F_c) / 2 \\ &= 0.5 (1 - k_c) \cdot F \\ &= k_s \cdot F \end{aligned} \quad (23)$$

where:  $k_s = 0.5 (1 - k_c)$

It should be noted that  $F_s$  could be calculated in the same way as  $F_c$  using the concept of the mean shear flow  $(q_s)_m$ .

These calculations are performed on the University IBM 1620 computer. The parameters are varied as follows:

1. Breadth/depth ratio i.e. B/D from 1.5 to 3.0, every 0.5.
2. Thickness ratio of effective deck plating/effective bottom plating i.e.  $t_D/t_B$  from 0.6 to 1.4, every 0.4.
3. Thickness ratio of effective bottom plating/side shell plating i.e.  $t_B/t_s$  from 0.6 to 1.4, every 0.4.
4. Thickness ratio of effective longitudinal bulkhead plating/side shell plating. i.e.  $t_c/t_s$  from 0.5 to 2.0, every 0.5.

For every condition of the above independent parameters, the programme computes the following:

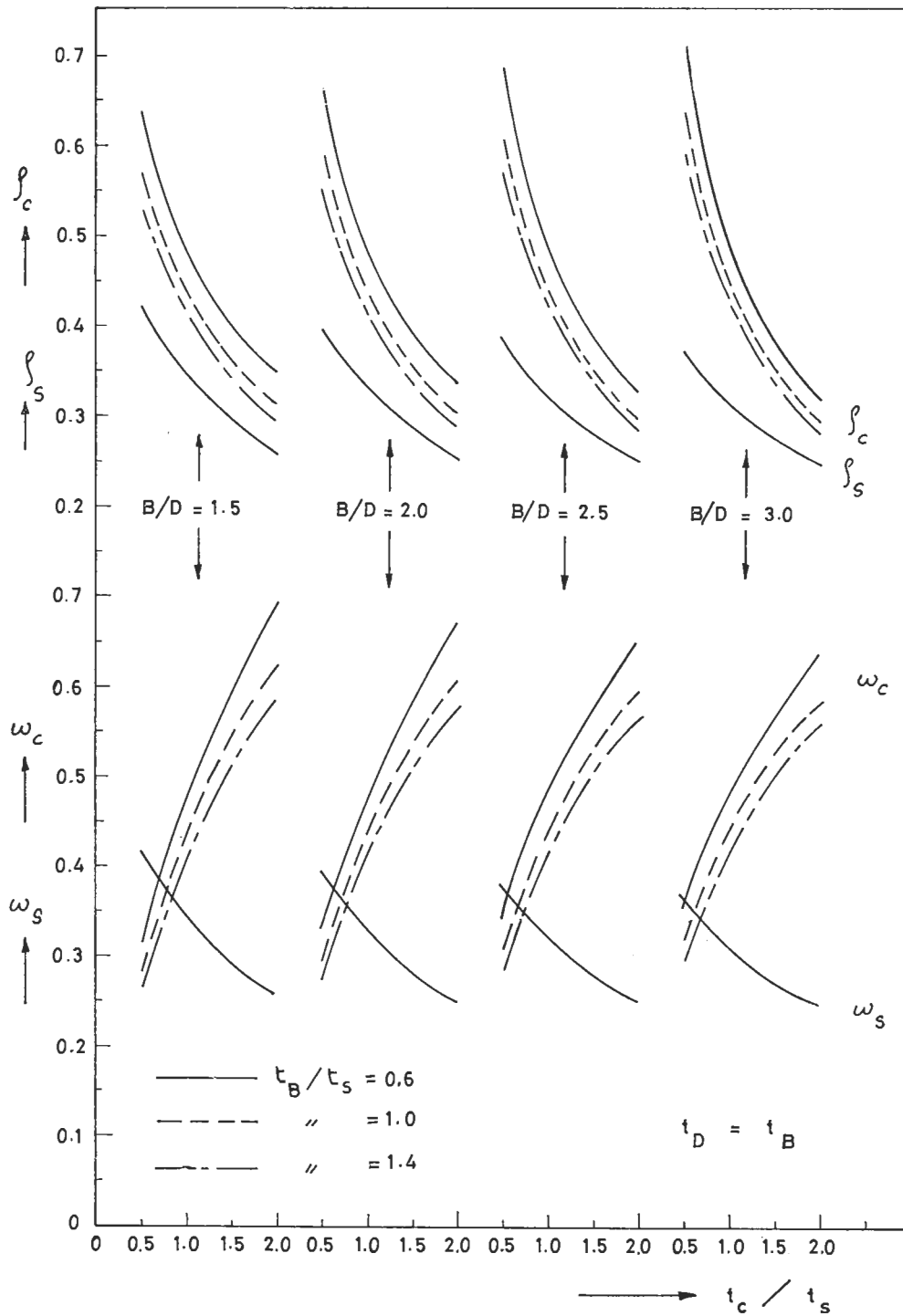


Fig. 2.

- i. A set of dependent parameters, as given in Appendix (I).
- ii. Maximum shear flow and shear stress at ship section neutral axis for both the centre line longitudinal bulkhead and side shell.
- iii. Shear loads carried by the longitudinal bulkhead and side shell.

*Results of calculations*

In these results the effect of varying each independent parameter, while keeping the other ship section parameters constant, on the maximum shear flow, maximum shear stress and on the shear loads carried by the side shell and longitudinal bulkheads are given. The results are presented in tabular and graphical form and are given in terms of both the

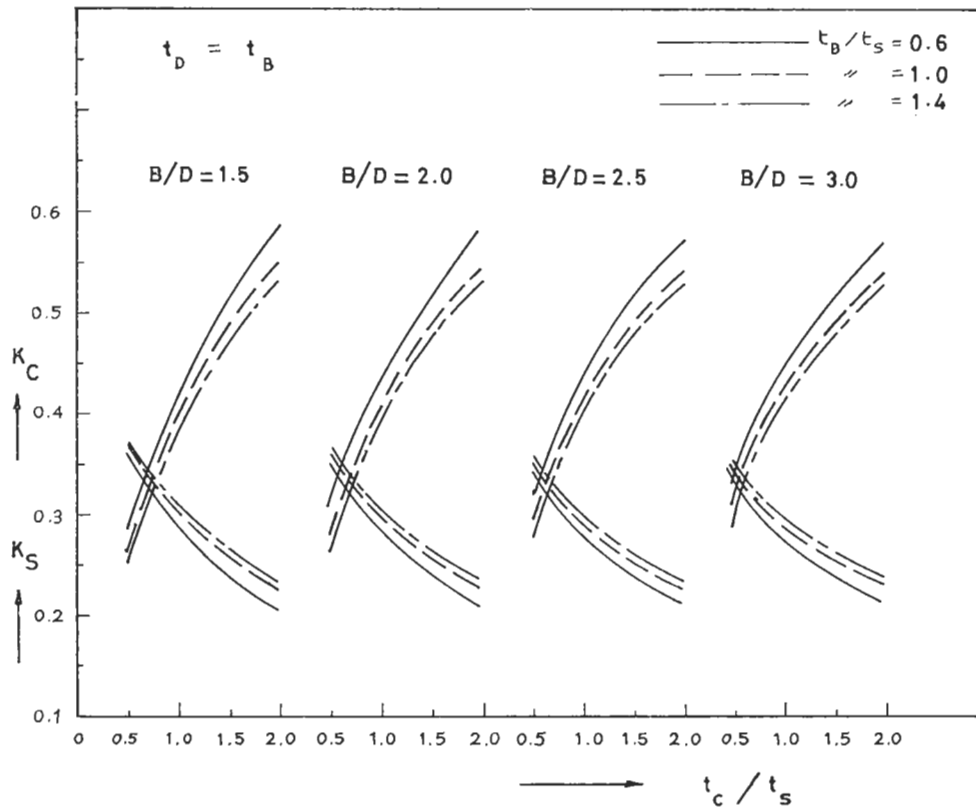


Fig. 3.

dependent and independent parameters. The percentage change in any calculated quantity is computed as follows:

$$\text{percentage change} = \frac{\text{final value} - \text{initial value}}{\text{initial value}} \times 100$$

The results of these calculations are summarised as follows:

1. Effect of B/D ratio:

The effect of increasing B/D ratio from 1.5 to 3.0 is shown in figs. (2, 3, 4) for the different values of  $t_B/t_S$  and  $t_C/t_S$ . For the special case, when  $t_D = t_B$ ,  $t_B/t_S = 1.4$ , the effect of B/D ratio for different values of  $t_C/t_S$  is given in the following table:

$t_C/t_S$	Percentage change			
	$q_C$ & $\eta_C$	$q_S$ & $\eta_S$	$F_C$	$F_S$
0.5	+11.0	-8.3	+15.0	-5.0
1.0	+2.6	-7.9	+6.5	-4.0
2.0	-5.2	-4.0	-0.9	+1.0

2. Effect of  $t_D/t_B$  ratio:

The effect of increasing  $t_D/t_B$  from 0.6 to 1.4 when  $B/D = 2.0$  and  $t_B/t_S = 1.4$  for different values of  $t_C/t_S$  is given in the following table:

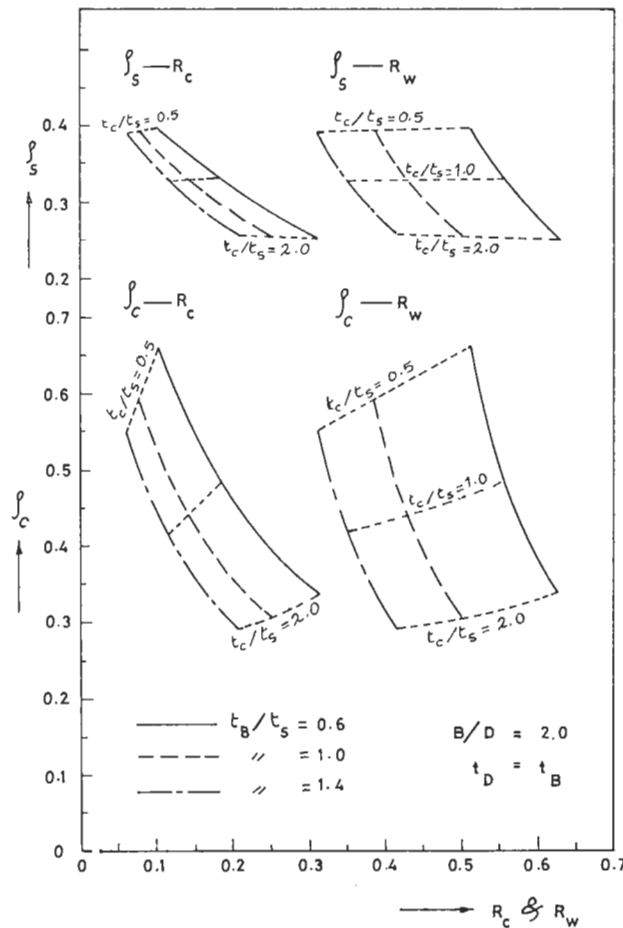


Fig. 4.

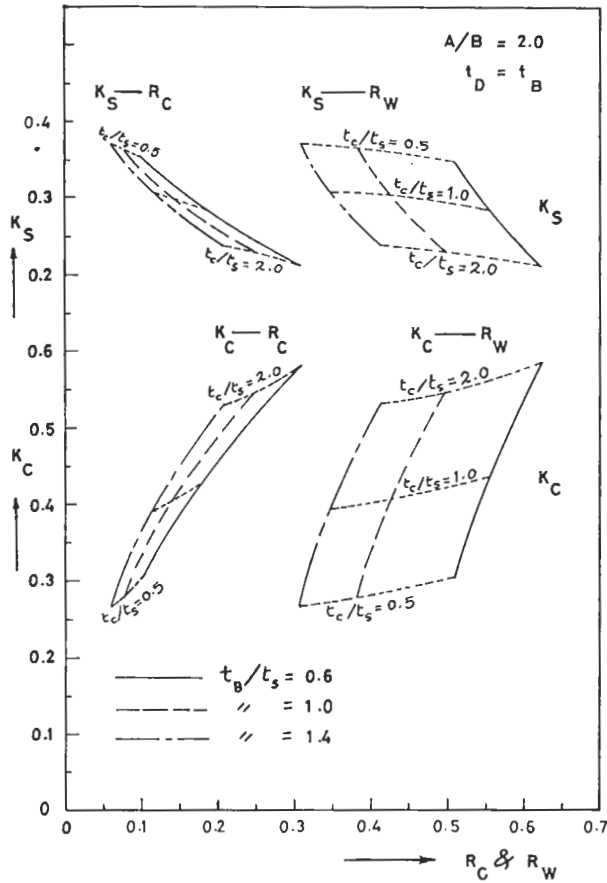


Fig. 5.

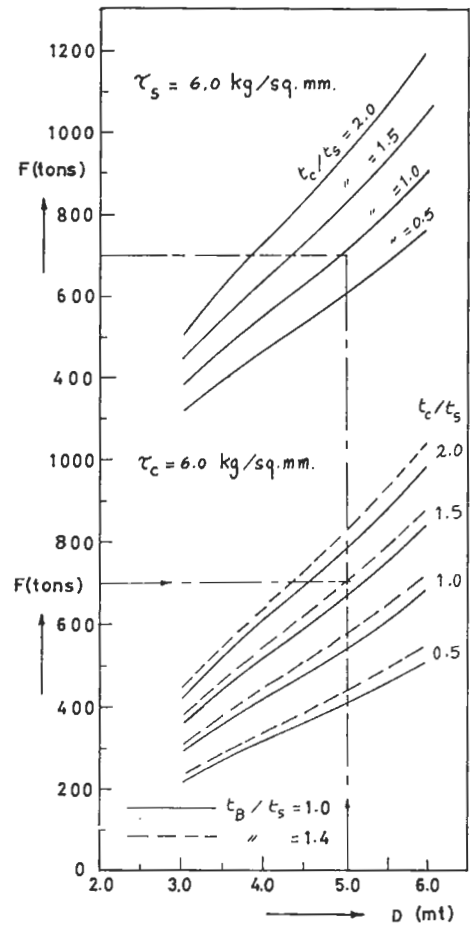


Fig. 6.

$t_c/t_s$	Percentage change			
	$q_c$ & $\eta_c$	$q_s$ & $\eta_s$	$F_c$	$F_s$
0.5	-8.7	-0.40	-7.2	+2.7
1.0	-7.0	-0.27	-4.8	+3.3
2.0	-6.6	+0.50	-4.0	+4.8

3. Effect of  $t_B/t_s$  ratio:

The effect of increasing  $t_B/t_s$  ratio from 0.6 to 1.4 is shown in figs. (2, 3, 4) for different values of  $B/D$ ,  $t_c/t_s$ . For the special case, when  $B/D = 3.0$ ,  $t_D = t_B$ , the effect of  $t_B/t_s$  ratio, for different values of  $t_c/t_s$  is given in the following table:

$t_c/t_s$	Percentage change									
	$q_c$		$\eta_c$		$q_s$ & $\eta_s$		$F_c$		$F_s$	
	B/D									
	2.0	3.0	2.0	3.0	2.0	3.0	2.0	3.0	2.0	3.0
0.5—1.0	+51.0	+44.0	-24.5	-28.0	-16.5	-16.5	+49.0	+42.2	-17.4	-17.0
1.0—2.0	+38.8	+32.0	-30.5	-34.0	-21.4	-19.5	+35.0	+29.5	-22.8	-20.0

$t_c/t_s$	Percentage change			
	$q_c$ & $\eta_c$	$q_s$ & $\eta_s$	$F_c$	$F_s$
0.5	-17.0	-1.40	-15.0	+7.5
1.0	-12.5	+0.84	-9.5	+7.7
2.0	-12.0	+2.50	-8.0	+10.7

4. Effect of  $t_c/t_s$  ratio:

It is shown that  $t_c/t_s$  ratio is the most effective parameter as it has the greatest influence on the magnitude and distribution of shear stresses in the structure. The results of increasing  $t_c/t_s$  from 0.5 to 2.0 are shown in figs. (2, 3, 4) for different values of  $B/D$  and  $t_B/t_s$  ratios. For the special cases when  $t_D = t_B$ ,  $t_B/t_s = 1.4$  and for two values of  $B/D$ , the effect of  $t_c/t_s$  is given in the following table:



5. Effect of  $R_W$  ratio:

The effect of  $R_W$ , when  $B/D = 2.0$ , is shown in figs. (5, 6). It is shown that increasing  $R_W$  by 20% has the following effects:

- a.  $\eta_c$  is reduced by more than 50%
- b.  $\eta_s$  is reduced by about 34%
- c.  $F_c$  is increased by about 90%
- d.  $F_s$  is reduced by about 66%

6. Effect of  $R_L$  ratio:

The effect of  $R_L$  when  $B/D = 2.0$ , is shown in figs. (5, 6). It is shown that increasing  $R_L$  by 100% gives the following results:

- a.  $\eta_c$  is reduced by about 31%
- b.  $\eta_s$  is reduced by about 19%
- c.  $F_c$  is increased by about 50%
- d.  $F_s$  is reduced by about 100%

*Analysis of Results*

For a certain value of the longitudinal vertical shear force, the effect of the different parameters on  $q_s$ ,  $\eta_s$ ,  $q_c$ ,  $\eta_c$ ,  $F_s$  and  $F_c$  are analysed and discussed. The analysis is intended mainly to examine the configuration of the ship section which brings the lowest shear stresses in both side shell plating and longitudinal bulkhead. Further, these analyses will make it possible to differentiate between the most and least effective parameters, regarding the shear carrying capacity of the ship hull girder.

a. *Effect of different parameters:*

- 1. Increasing  $B/D$  ratio always tends to reduce  $q_s$ ,  $\eta_s$  &  $F_s$  when  $t_c/t_s$  ratio is low and has a negligible effect when  $t_c/t_s$  ratio is high.
- 2. Increasing  $B/D$  ratio always tends to increase/decrease  $q_c$ ,  $\eta_c$  &  $F_c$  when  $t_c/t_s$  ratio is low/high.
- 3. Increasing  $t_D/t_B$  ratio always tends to reduce  $q_c$ ,  $\eta_c$ ,  $F_c$  &  $F_s$
- 4.  $t_D/t_B$  ratio has a negligible effect on  $q_s$  and  $\eta_s$
- 5. Increasing  $t_B/t_s$  ratio always tends to reduce  $q_c$ ,  $\eta_c$  and  $F_c$  and increases  $F_s$
- 6.  $t_B/t_s$  ratio has, in general, a small effect on  $q_s$  and  $\eta_s$
- 7. Increasing  $t_c/t_s$  ratio always tends to increase  $q_c$  and  $F_c$  and reduce,  $\eta_c$ ,  $q_s$ ,  $\eta_s$  and  $F_s$
- 8. Increasing  $R_L$  and/or  $R_W$  always tends to increase  $F_c$  and reduce  $\eta_s$ ,  $\eta_c$  and  $F_s$

Consequently, in order to keep down the max-

imum shear stress in both side shell plating and centre line longitudinal bulkhead, the following parameters should be varied as follows:

- i.  $t_D/t_B$  and  $t_B/t_s$  ratios should be increased
- ii.  $B/D$  ratio should be decreased
- iii.  $t_c/t_s$  should be increased
- iv.  $R_L$  and/or  $R_W$  should be increased.

It should be realised that  $B/D$ ,  $t_D/t_B$  and  $t_B/t_s$  ratios have a minor effect on  $\eta_c$  &  $\eta_s$  whereas  $t_c/t_s$ ,  $R_L$  and/or  $R_W$  ratios have the major effect on the magnitude of  $\eta_c$  &  $\eta_s$

b. *Maximum Allowable Shear Force*

The results of this investigation could also be used to calculate the maximum shear force which will not induce shear stresses in the side shell plating and/or the centre line longitudinal bulkhead greater than a maximum allowable value.

The maximum shear stress in the side shell plating and centre line longitudinal bulkhead due to a longitudinal vertical shear force are given by:

$$\eta_s = \zeta_s \cdot \frac{F}{D \bar{t}_s}$$

and 
$$\eta_c = \zeta_c \cdot \frac{F}{D \bar{t}_s}$$

The corresponding shear force is therefore given by:

$$F = \frac{\eta_s \cdot D \bar{t}_s}{\zeta_s}$$

or 
$$= \frac{\eta_c \cdot D \bar{t}_s}{\zeta_c}$$

If  $\eta_s \leq \eta_a$

and  $\eta_c \leq \eta_a$

where  $\eta_a$  = maximum allowable shear stress and varies between 6.0 and 7.0 kg/sq. mm.

Hence; 
$$F_{\max} \leq \frac{\eta_a \cdot D \bar{t}_s}{\zeta_s} \dots (a)$$

or 
$$\leq \frac{\eta_a \cdot D \bar{t}_s}{\zeta_c} \dots (b)$$

where:  $F_{\max}$  = maximum value of shear force obtained from longitudinal strength calculation.

The calculation of  $F_{\max}$  from either (a) or (b) can only be performed when the thickness of side plating i.e.  $\bar{t}_s$  is known.

Using G.L. Rules (1963) for small tankers (length  $\leq 80$  m.), the thickness of side shell plating when longitudinally framed is given by:

$$s = \frac{a_1}{110} \sqrt{\frac{0.85 T + 0.025 L}{2.2 + 0.01 L}}$$

where  $s$  = thickness of side shell plating in mm.  
 $a_1$  = spacing of longitudinals in mm.  
 $L$  = length of ship in m.  
 $T$  = draft of ship in m.

It is to be noted that:  $a_1 < 0.9 (2L + 500)$   
 and minimum thickness  $\geq 0.04 L + 5.3$  mm.

Consequently, using the above expressions, a relationship between the ship depth  $D$  and the product  $D \bar{t}_s$  could be determined either analytically or in numerical form.

Substituting  $D \bar{t}_s$  into expressions (a) and (b), we get:

$$F_{\max} = f_s (\eta_a, \zeta_s, D) \dots (c)$$

$$\text{or } F_{\max} = f_L (\eta_a, \zeta_c, D) \dots (d)$$

where:  $f_s$  and  $f_c$  : are the functions relating  $\eta_a$ ,  $\zeta$ ,  $D$  with  $F_{\max}$ .

Expressions (c) and (d) could be represented graphically as shown in Fig. (6). From these curves, at any depth  $D$  of a ship, the maximum allowable shear force which will not induce shear stress, in either the side shell plating or the centre line longitudinal bulkhead, greater than a maximum allowable value, could be determined. Alternatively, for any depth  $D$  and maximum shear force  $F$ , the different ratios of  $t_B/t_s$  and  $t_c/t_s$  which will not induce shear stresses greater than  $\eta_a$  could be determined.

Several curves could be obtained for the different conditions of  $B/D$ ,  $t_D/t_B$ ,  $t_B/t_s$  and  $t_c/t_s$  ratios for both transverse and longitudinal framing systems.

As an example, for a longitudinally framed tanker having:

$$D = 5.0 \text{ m.} \quad B/D = 2.0$$

$$\eta_a = 6.0 \text{ kg/sq. mm.}$$

and assuming that the maximum shear force obtained from longitudinal strength calculations is 700 tons.

In order to satisfy the condition:

$$\eta_s \leq \eta_a \dots (e)$$

$$\text{and } \eta_c \leq \eta_a \dots (f)$$

the following section parameters are obtained from Fig. (6).

i. In order to satisfy condition (e), we have:

$$t_c/t_s \geq 1.0$$

ii. In order to satisfy condition (f), we have:

$$t_c/t_s \geq 1.7 \text{ when } t_B/t_s = 1.0$$

$$\text{or } \geq 1.5 \text{ when } t_B/t_s = 1.4$$

Since the minimum value of  $\bar{t}_s$  is determined from the C.L. Rules, the minimum thickness of the centre line longitudinal bulkhead which will satisfy conditions (e) and (f) can be calculated. This type of study will be most useful in the case of large tankers having twin or three longitudinal bulkheads since the efficient use of the bulkhead material will certainly lead to an appreciable saving in the total steel weight.

### Conclusions

Longitudinal bulkheads, apart from dividing the hull into water-tight compartments, are important structural elements of the main hull girder. The contribution of these bulkheads to longitudinal bending and shear forces depends mainly on their resistance to local buckling.

However, when these bulkheads are adequately stiffened to prevent local buckling, it is possible to increase their contribution to the basic loads of the main hull girder. As a result, the thickness of side shell plating could be reduced since it will be subjected to a reduced bending and shear loads. The reduced thickness should be checked for instability resistance, excessive deflections and high stresses due to normal hydrostatic pressure and hydrodynamic loading.

It is also concluded that  $t_c/t_s$  and  $R_L$  or  $R_W$  are the most important parameters which affect the maximum shear stress in the side shell plating and centre line longitudinal bulkhead in addition to affecting the contribution of the latter to the shear carrying capacity of the main hull girder.

Further, it is shown that it is possible to obtain from a series of curves, the conditions of the ship section which, under a maximum longitudinal vertical shear force, will not induce shear stress in the side shell plating or centre line longitudinal bulkhead greater than a maximum allowable value.

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Appendix (I)

a) Independent Parameters

- i.  $t_D/t_s = x$
- ii.  $t_B/t_s = z$
- iii.  $t_c/t_s = y$
- iv.  $B/D = \gamma$

where  $t$  is the effective thickness.

b) Dependent Parameters

- i. Total sectional area of all longitudinal material =  $A_T$

$A_T$  = sectional area of deck, bottom, longitudinal bulkhead and side shell plating

$$A_T = [2 + y + \gamma z(1+x)] \cdot D t_s$$

- ii. Sectional area of shear carrying members =  $A_w$

$A_w$  = sectional area of longitudinal bulkhead and side shell plating

$$A_w = [2 + y] \cdot D t_s$$

- iii. Sectional area of centreline longitudinal bulkhead =  $A_c$

$$A_c = y \cdot D t_s$$

$$iv. A_w/A_T = \frac{2 + y}{2 + y + \gamma z(1+x)} = R_w$$

$$v. \frac{A_c}{A_T} = \frac{y}{2 + y + \gamma z(1+x)} = R_c$$

- vi. Position of neutral axis from base line.

This is defined by the normalised distance  $\beta$ , see fig. (1) and is given by:

$$= \frac{(1 + 0.5y) + \gamma x z}{2 + y + \gamma z(1+x)}$$

- vii. Second moment of area  $I$ , ( $I = \psi D^3 t_s$ )

This is defined by the non-dimensional coefficient  $\psi$  which is given by:

$$\psi = (2 + y) \left[ \frac{1}{12} + (0.5 - \beta)^2 \right] + \gamma z [\beta^2 + x(1 - \beta)^2]$$

APPENDIX II

SHEAR FLOW DISTRIBUTION

MEMBER OR POINT	$A \bar{y} / D^2 t_s$	$\omega_i = q_i / \frac{F}{D}$	$\bar{q} / \frac{F}{D}$	$\frac{\Delta s}{t} / \frac{D}{t_s}$	$\bar{q} \frac{\Delta s}{t} / \frac{F}{t_s}$	$(q_i)_r$
C	-	-				$-2(q)_c$
AC	$y(1-\beta)^2/2$	$y(1-\beta)^2/2$	$y(1-\beta)^2/6$	$(1-\beta)/y$	$(1-\beta)^3/6$	$q_{AC} - (q)_c$
AB	$\gamma x z(1-\beta)/2$	$y(1-\beta)^2/4$	$3y(1-\beta)^2/8 + \gamma z x(1-\beta)/4$	$\gamma/2 x z$	$3\gamma y(1-\beta)^2/16 x z + \gamma^2(1-\beta)/8$	$q_{AB} - (q)_c$
BA		$y(1-\beta)^2/4 + \gamma x z(1-\beta)/2$				$q_{BA} - (q)_c$
BS	$(1-\beta)^2/2$		$(\frac{y}{4} + \frac{1}{3})(1-\beta)^2 + \gamma x z(1-\beta)/2$	$1-\beta$	$(\frac{y}{4} + \frac{1}{3})(1-\beta)^3 + \gamma x z(1-\beta)^2/2$	$q_{BS} - (q)_c$
SB		$(\frac{y}{2} + 1)(1-\beta)^2/2 + \gamma x z(1-\beta)/2$				$q_{SB} - (q)_c$
EC	$y\beta^2/2$	$y\beta^2/2$	$y\beta^2/6$	$\beta/y$	$\beta^3/6$	$q_{EC} - (q)_c$
ED	$\gamma z\beta/2$	$y\beta^2/4$	$3y\beta^2/8 + \gamma z\beta/4$	$\gamma/2 z$	$3\gamma y\beta^2/16 z + \gamma^2\beta/8$	$q_{ED} - (q)_c$
DE		$\gamma z\beta/2 + y\beta^2/4$				$q_{DE} - (q)_c$
DS	$0.5\beta^2$		$(\frac{y}{4} + \frac{1}{3})\beta^2 + \gamma z\beta/2$	$\beta$	$(\frac{y}{4} + \frac{1}{3})\beta^3 + \gamma z\beta^2/2$	$q_{DS} - (q)_c$
SD		$(\frac{y}{2} + 1)\beta^2/2 + \gamma z\beta/2$				$q_{SD} - (q)_c$

$$(q)_c = \frac{\int_A q \frac{\Delta s}{t}}{\int_A \frac{\Delta s}{t} + 2 \int_E \frac{\Delta s}{t}} = \left\{ \frac{(\frac{y}{4} + \frac{1}{3})[\beta^3 + (1-\beta)^3] + 3\gamma y[\beta^2 + (1-\beta)^2/x]/16z + \gamma z[\beta^2 + x(1-\beta)^2]/2 + \gamma^2\beta}{\gamma(1+x)/2z + 2/y + 1.0} \right\} \frac{F}{I} \cdot D^2 t_s$$

Substituting for  $I = \psi D^3 t_s$

and  $(q)_c$  for the term between brackets

$$\therefore (q)_c = (\omega)_c \cdot \frac{F}{D}$$

where  $(\omega)_c = (q)_c / \psi$